

Comparative thermal analysis of an advanced ceramic-coated piston in a spark ignition engine**

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This study deals with the steady-state temperature variation in the partially ceramic-coated Al–Si piston of a petrol engine investigated using ANSYS Workbench finite-element modeling software. The ceramic material used for coating the piston crown was lanthanum cerate ($\text{La}_2\text{Ce}_2\text{O}_7$). Coating width and thickness effects were explored and compared with an uncoated piston. Convection boundary conditions were considered. The coated surface temperature is greater than that of the uncoated piston and increases with increasing coating thickness, while normal stress decreased.

Keywords: ceramic coating, finite element modeling, internal combustion engine, lanthanum cerate, thermal barrier coating

1. Introduction

Nowadays, improvement of total efficiency and performance of spark ignition (SI) engine components, internal combustion (IC) engine components and gas turbines are achieved using thermal barrier coatings (TBCs) [1–5]. They have also been used in adiabatic engines, lowering heat rejection and failure fatigue of the inner metallic surfaces while also reducing harmful emissions [6–10]. A TBC consists of several layers, viz. successively a substrate, an intermediate bond coat layer and a top ceramic coating consisting of lanthanum cerate [11, 12]. As well as hindering heat transfer (heat flux), TBCs also protect the piston from corrosive attack [13–16]. Better utilization of the energy carried by the exhaust increases the work done inside the cylinder [8, 15, 17].

Hence, in SI engines the use of TBCs leads to improvements in performance and efficiency and, furthermore, a decrease of harmful exhaust emissions from the engine [18–22]. In the present analysis, a partial ceramic coating was applied to the piston of a SI engine. In previous research [19], hydrocarbon (HC) emissions beginning from a cold start of a standard (uncoated)

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engine were compared with those from engines having partially ceramic-coated pistons. Significant reductions of CO and HC emissions were observed with the coated pistons. A temperature increase of 100 °C at the ceramic-coated surface (coating thickness 0.5 mm) was observed [18, 19].

The topmost ceramic layer decreases the internal stress that arises within the substrate during thermal shock; the coating thickness has a direct effect on the thermal stress. The modulus of elasticity, thermal conductivity and thermal expansion coefficient of the materials are the main factors on which the robustness of a ceramic coating depends during thermal shock [21, 23, 24]. Calculation of the temperature distribution within the piston is important for controlling the deformations and stresses. Thermomechanical analysis contributes to the design of the piston of an SI engine before the actual manufacturing and testing.

2. Construction of partially coated piston

The piston was made from Al–Si alloy. Material corresponding to the thickness and width of the coating was removed from the piston's surface at its circumference. A NiCrAl bond coat was then applied to form a layer of thickness $s = 0.15$ mm, onto which a lanthanum cerate ($\text{La}_2\text{Ce}_2\text{O}_7$) coating (thickness t) was applied (at the crown of the piston) by plasma spraying, as shown in Fig. 1.

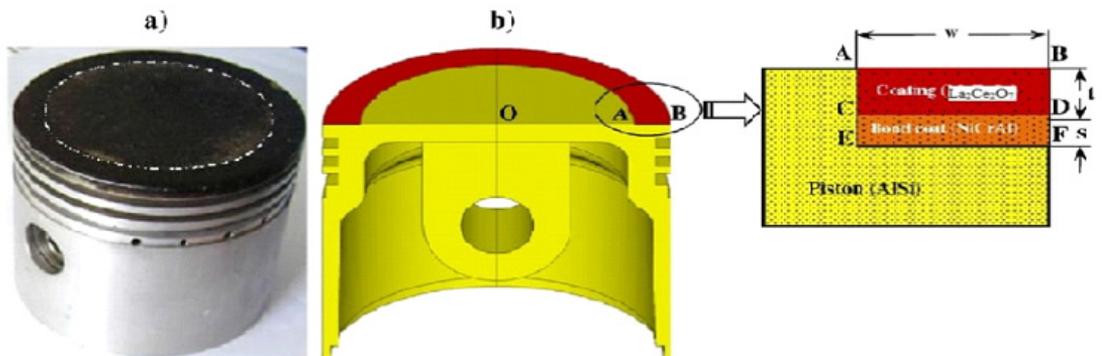


Figure 1. Model for FEM analysis. (a) Photograph of partially ceramic-coated aluminium alloy piston with cast iron rings; (b) cutaway view (half part) of the piston [22]. Inset: width and thicknesses of the TBC applied to the crown of the piston.

3. Thermal analysis

The effects for a petrol engine of the TBC were investigated via steady-state thermomechanical analysis in 3D using ANSYS finite element modeling (FEM) software; a liquid-cooled, single-cylinder SI engine piston was modeled. Widths w and coating thicknesses of the three layers (i.e., the top coat forming the surface, the bond coat and the substrate) are given in Table 1 (cf. Fig. 1b, inset). Specifications of the engine are given in Table 2. Table 3 gives the thermomechanical constants of the different materials used in this analysis. Fig. 2 shows the finite element mesh used for the modeling.

Table 1. Layer thicknesses and widths (cf. inset to Fig. 1b).

t / mm	s / mm	$t + s$ / mm	w / mm
0.15, 0.25, 0.35,	0.15	0.3, 0.4, 0.5, 0.6,	9.2
0.45, 0.55, 0.65,		0.7, 0.8, 1.0, 1.2,	
0.75, 0.85, 0.95,		1.4, 1.6	
1.05, 1.15, 1.25,			
1.35, 1.45			
0.35	0.15	0.5	5.2, 7.2, 9.2, 11.2

Table 2. Specifications of the SI engine modeled in this work.

Bore	85.95 mm
Stroke	83.45 mm
Displacement	470.08 mm ³
Compression Ratio	7.4
Power (at 3000 rpm)	45 kW
Maximum speed	3400 rpm

Table 3. Physical constants of the materials used in this work [17, 18, 22, 26–29].

Material (function)	Density / kg m ⁻³	Specific heat / J kg ⁻¹ K ⁻¹	Modulus of elasticity / Gpa	Poisson's ratio	Thermal conductivity / W m ⁻¹ K ⁻¹	Thermal expansion coefficient / K ⁻¹ × 10 ⁻⁶
Cast iron (piston rings)	7300	460	200	0.30	16	10
AlSi (piston)	2700	960	69	0.33	155	21
NiCrAl (bond coat layer)	7870	764	90	0.27	16.1	12
La ₂ Ce ₂ O ₇ (ceramic top coating)	6290	430	25	0.29	0.6	12.9

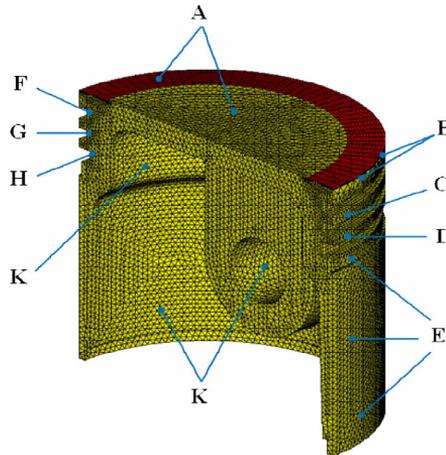


Figure 2. FEM mesh of the cylinder (the letters refer to Table 4).

Convection was selected as the heat transfer mechanism. Heat transfer in SI engines is complex due to the following reasons:

- effects on piston motion arising through neglect of heat transfer;
- possible twisting of the rings (should not occur);
- the piston crown and the rings should be sufficiently oiled to avoid a cavity [14].

Hohenberg proposed the following equation for the instantaneous convective heat transfer coefficient [25]:

$$h_{\text{gas}}(T) = \alpha V_C(T)^{-0.06} P(T)^{-0.4} (S_p + b)^{0.8} \quad (1)$$

where V_C , P and T are the instantaneous cylinder volume, pressure and temperature respectively, and S_p is the mean piston speed; α and b are constants. Table 4 gives the boundary conditions applied.

Table 4. Boundary conditions of the piston [22].

Piston region (cf. Fig. 2)	$h_{\text{gas}} / \text{W m}^{-2} \text{K}^{-1}$	$T / ^\circ\text{C}$
A	600	650
B	350	500
C	300	180
D	400	170
E	400	110
F	400	200
G	400	180
H	400	170
K	1500	95

4. Results and discussion

Fig. 3 shows the maximum temperature obtained at the piston crown of both the uncoated piston and coated piston. Maximum temperature of 314.13 °C and minimum temperature of 107.76 °C were obtained for the uncoated piston. Maximum temperature for the partially ceramic-coated piston is obtained with $t + s = 0.5$ mm and $w = 11.2$ mm. The low thermal conductivity of the ceramic material with respect to the aluminium alloy substrate is clearly effective in augmenting the maximum temperature of the piston (preliminary work showed that lanthanum cerate was a better ceramic material than magnesium zirconate due to its lower thermal conductivity). The temperature at the top ceramic-coated surface increases with increase in thickness and width. Radial distance versus ceramic-coated top surface temperature distributions are shown in Fig. 4.

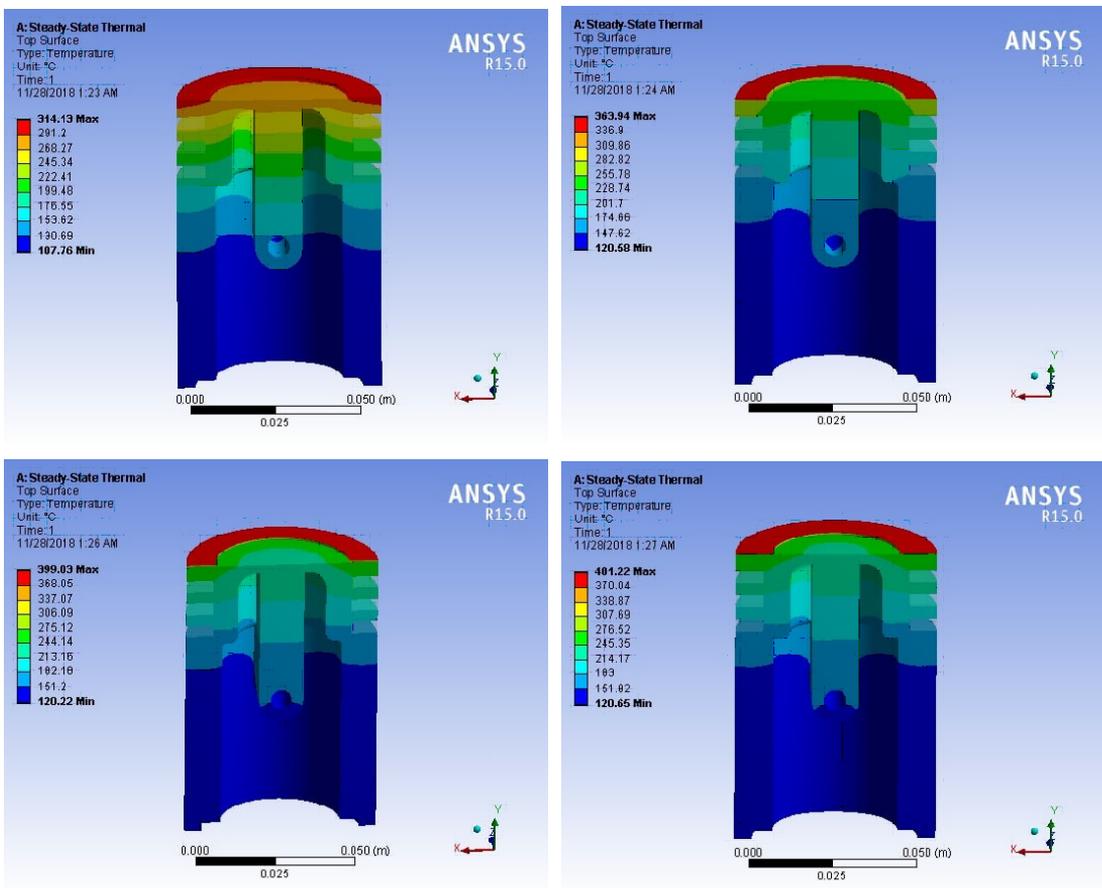


Figure 3. Temperature/°C contours obtained by FEM for the SI engine piston. Top left, uncoated; top right, total thickness of coating 0.3 mm and width 9.2 mm; bottom left, total thickness 0.5 mm and width 9.2 mm; bottom right, total thickness 0.5 mm and width 11.2 mm.

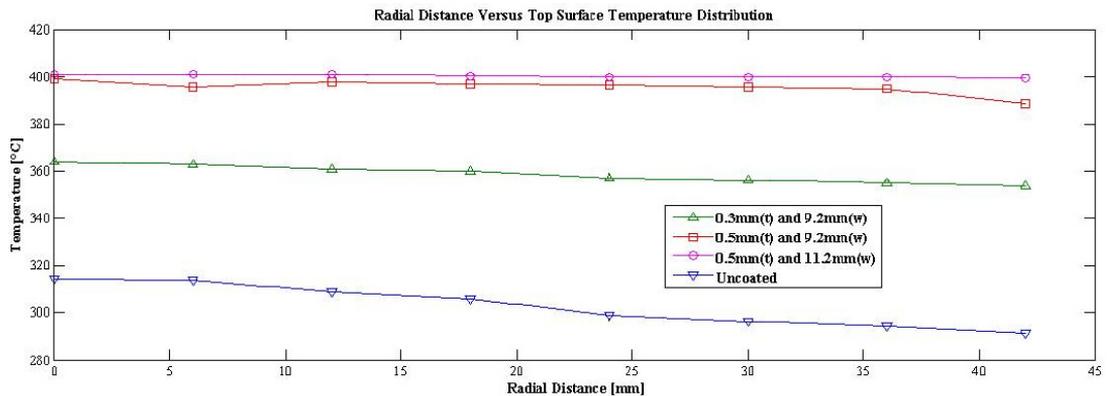


Figure 4. Temperature on the top surface as a function of radial distance.

5. Conclusions

FEM of an SI engine piston clearly shows that the piston's top surface temperature depends upon both ceramic coating thickness and width. The greatest increase in the maximum temperature of the top surface of the piston is almost 28% compared with the uncoated piston, while the piston's substrate temperature decreases, which together have a positive impact on the performance and efficiency of the engine.

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